

Aircraft Wing Vibrations During Landing

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Introduction

- ▶ Aeroplane hits the runway and landing gear causes shock
- ▶ Model wing as thin elastic beam
- ▶ Assume stationary
- ▶ At $t = 0$, gear acts

$$M \propto k \implies m = bk \quad (b > 0)$$

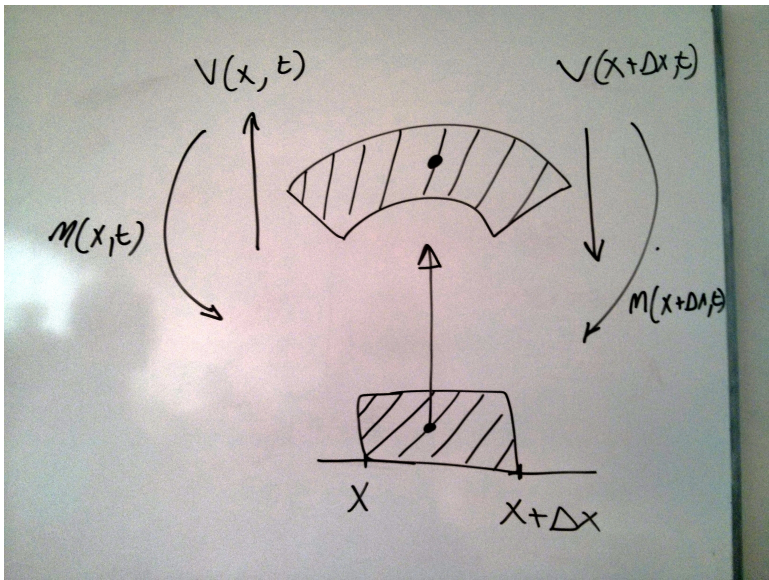


Figure: An elemental section of the 1 dimensional beam.

Rotational Equilibrium

In equilibrium, it is known that

$$V(x - \frac{\Delta x}{2}, t)(-\frac{\Delta x}{2}) - V(x + \frac{\Delta x}{2}, t)(\frac{\Delta x}{2}) + M(x - \frac{\Delta x}{2}, t) - M(x + \frac{\Delta x}{2}, t) = 0$$

$$(V(x, t) - \frac{\partial V}{\partial x} \frac{\Delta x}{2})(-\frac{\Delta x}{2}) - (V(x, t) + \frac{\partial V}{\partial x} \frac{\Delta x}{2})(\frac{\Delta x}{2}) + M(x, t) - \frac{\partial M}{\partial x} (\frac{\Delta x}{2}) = 0$$

$$V(x, t)\Delta x - \frac{\partial M}{\partial x} \Delta x = 0$$

$$V(x, t) = \frac{\partial M}{\partial x} \quad (1)$$

Derivation of curvature and bending moment relationship

The curvature for a vector valued function is defined as,

$$\kappa := \left| \frac{d\mathbf{T}}{ds} \right|.$$

But for a function embedded in a plane with graph $y = f(x)$,

$$\kappa = \frac{|y''|}{(1 + y')^{3/2}}$$

It is assumed that slopes are small compared with unity, thus the curvature can be approximated as,

$$\kappa \approx \frac{d^2y}{dx^2} \quad (2)$$

As assumed

$$M = b\kappa = b \frac{\partial^2 u}{\partial x^2} \quad (3)$$

Kinematic equations

$$\mathbf{F}_{net} = m\mathbf{a}$$

$$V(x - \frac{\Delta x}{2}, t) - V(x + \frac{\Delta x}{2}, t) = \rho \frac{\partial^2 u}{\partial t^2} \Delta x$$

$$V(x, t) - \frac{\partial V}{\partial x} \frac{\Delta x}{2} - V(x, t) + \frac{\partial V}{\partial x} \frac{\Delta x}{2} = \rho \frac{\partial^2 u}{\partial t^2} \Delta x$$

$$-\frac{\partial V}{\partial x} = \rho \frac{\partial^2 u}{\partial x^2 t}$$

$$\text{where } V = \frac{\partial M}{\partial x}$$

$$-\frac{\partial^2 M}{\partial x^2} = \rho \frac{\partial^2 u}{\partial t^2}$$

$$M = b\kappa$$

$$\implies -b \frac{\partial^2 \kappa}{\partial x^2} = \rho \frac{\partial^2 u}{\partial t^2}$$

Using the relationship in eq(2), it can be concluded that

$$\frac{\partial^4 u}{\partial x^4} = \left(-\frac{\rho}{b}\right) \frac{\partial^2 u}{\partial t^2} \quad (4)$$

Which is generally known as **Euler-Bernoulli beam equation**.

It is now assumed that there is a gravitational force $\rho\Delta xg$ acting on the infinitesimal element as considered in figure(2), when the forces are equated,

$$\begin{aligned}V\left(x - \frac{\Delta x}{2}, t\right) - V\left(x + \frac{\Delta x}{2}, t\right) - \rho\Delta xg &= \rho\Delta x\frac{\partial^2 u}{\partial t^2} \\ \implies -\frac{\partial V}{\partial x}\Delta x &= \rho\Delta x\left(g + \frac{\partial^2 u}{\partial t^2}\right) \\ \frac{\partial V}{\partial x} &= \rho\Delta x\left(g + \frac{\partial^2 u}{\partial t^2}\right).\end{aligned}$$

Using eqs(1,2), the resulting equation is,

$$-b\frac{\partial^4 u}{\partial t^4} = \rho\left(g + \frac{\partial^2 u}{\partial t^2}\right)$$

Static solution

The **time-independent** static solution to the **Euler-Bernoulli beam equation** and the equation now becomes a *4th* order ODE given by,

$$\frac{d^4 u}{dx^4} = -\frac{\rho g}{b}$$

and the boundary conditions are given by,

$$u(0) = 0$$

$$u'(0) = 0$$

$$u''(L) = 0$$

$$u'''(L) = 0.$$

The solution to this IVP is given by,

$$u(x) = \frac{3g}{EL^3} \left(-\frac{1}{24}x^4 + \frac{L}{6}x^3 - \frac{L^2}{4}x^2 \right)$$

Plot

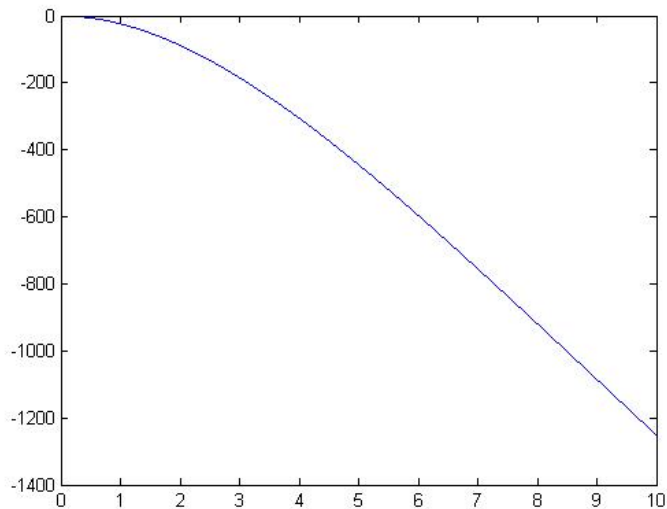


Figure: Non scaled plot of solution

Full Solution

We have

$$-b \frac{\partial^4 u}{\partial t^4} = \rho \left(g + \frac{\partial^2 u}{\partial t^2} \right)$$

$$u(0, t) = 0$$

$$u_x(0, t) = 0$$

$$u_{xx}(L, t) = 0$$

$$u_{xxx}(L, t) = 0$$

$$u(x, 0) = 0$$

$$u_t(x, 0) = 0$$

Using the separation of variables technique it we assume that the solution is of the form

$$u(x, t) = X(x)T(t) \quad (5)$$

We find that

$$u(x, t) = \sum_{n=1}^{\infty} A_n [[\cosh(\beta_n x) - \cos(\beta_n x)] - \frac{\omega}{\gamma} [\sinh(\beta_n x) - \sin(\beta_n x)]] \sin(\alpha_n t)$$

$$\frac{\partial^2 u}{\partial t^2} = -\alpha^2 \frac{\partial^4 u}{\partial x^4} \quad \text{--- (1)}$$

$$\rightarrow u(x,t) = a(k) \exp[ikx + \lambda(k)t] \quad \begin{matrix} (k \in \mathbb{R}) \\ t \geq 0 \end{matrix}$$

Substitute in (1),

$$\lambda^2(k) = -\alpha^2 k^4$$

$$\lambda^2(k) + \alpha^2 k^4 = 0 \Rightarrow \lambda(k) = \pm i\alpha k^2$$

$$\text{let } \lambda(k) = \pm i\alpha k^2 \quad \text{--- (2)}$$

\rightarrow The F.T of $u(x,t)$ be $H(\lambda,t)$

$$H(\lambda,t) = \frac{1}{\int_{-\infty}^{\infty} e^{-ikx}}$$

This is a very classical approach to solve the 1-d beam equation, however it does not reveal the boundary conditions and the physical nature of the problem.

QUESTIONS?